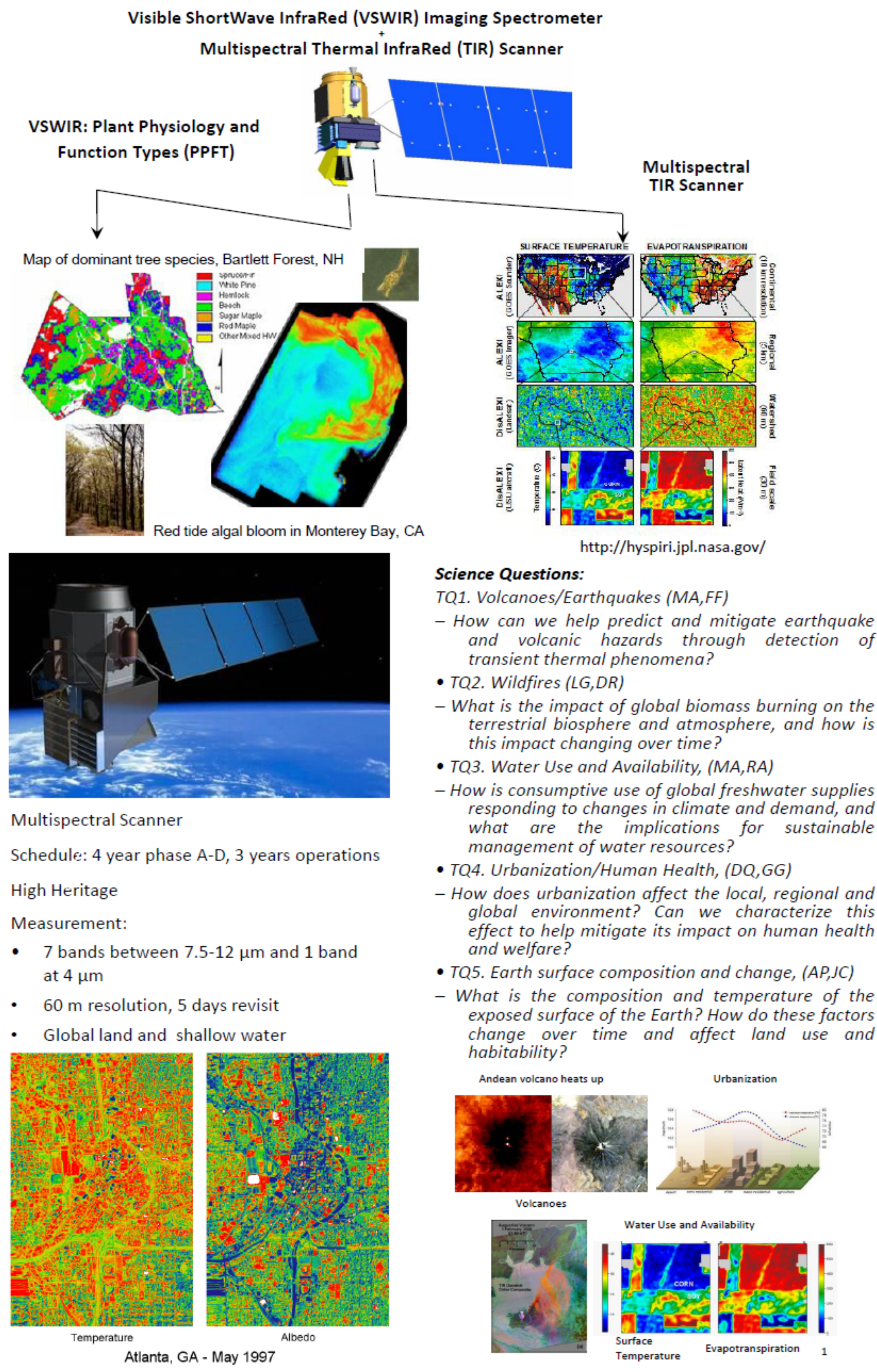


Prototype HypsIRI Thermal Infrared Radiometer (PHyTIR): status and test plan

William R. Johnson, Simon J. Hook (P.I.), Bjorn Eng, Marc C. Foote, Bruno M. Jau, Chris Paine, Robert Smythe & Thomas Werne

HypsIRI Background

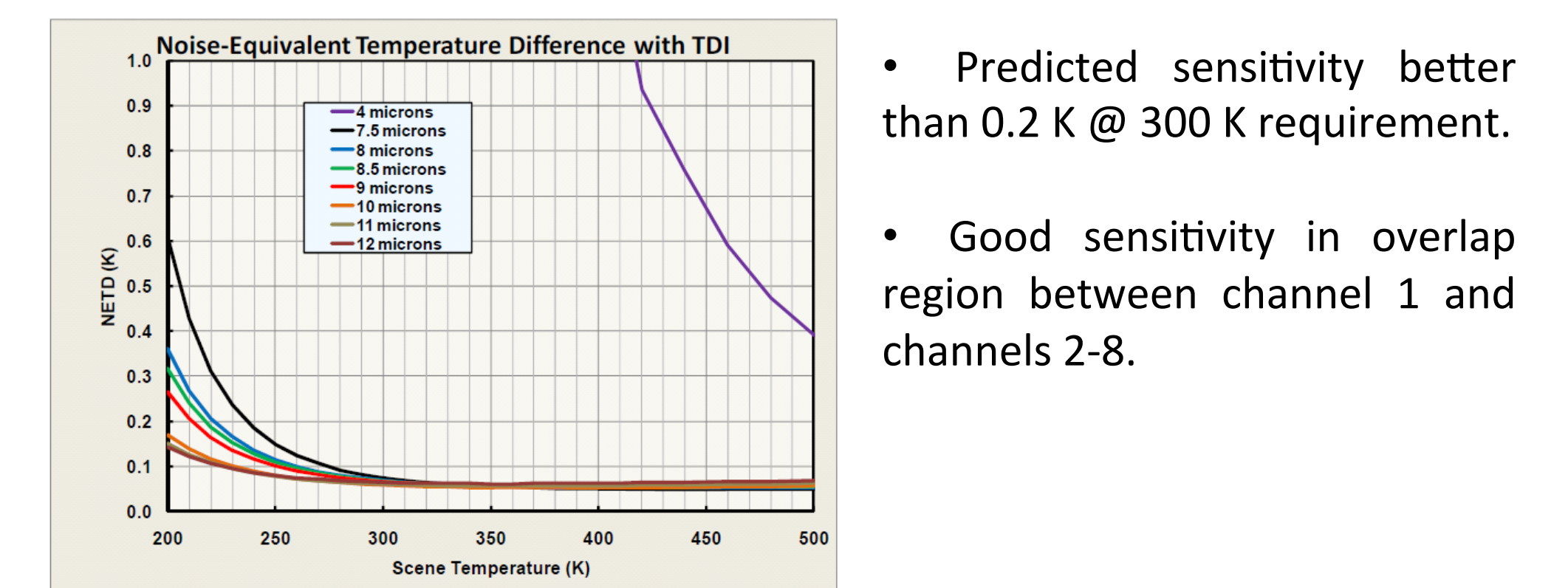


PHyTIR Rational for HypsIRI-TIR

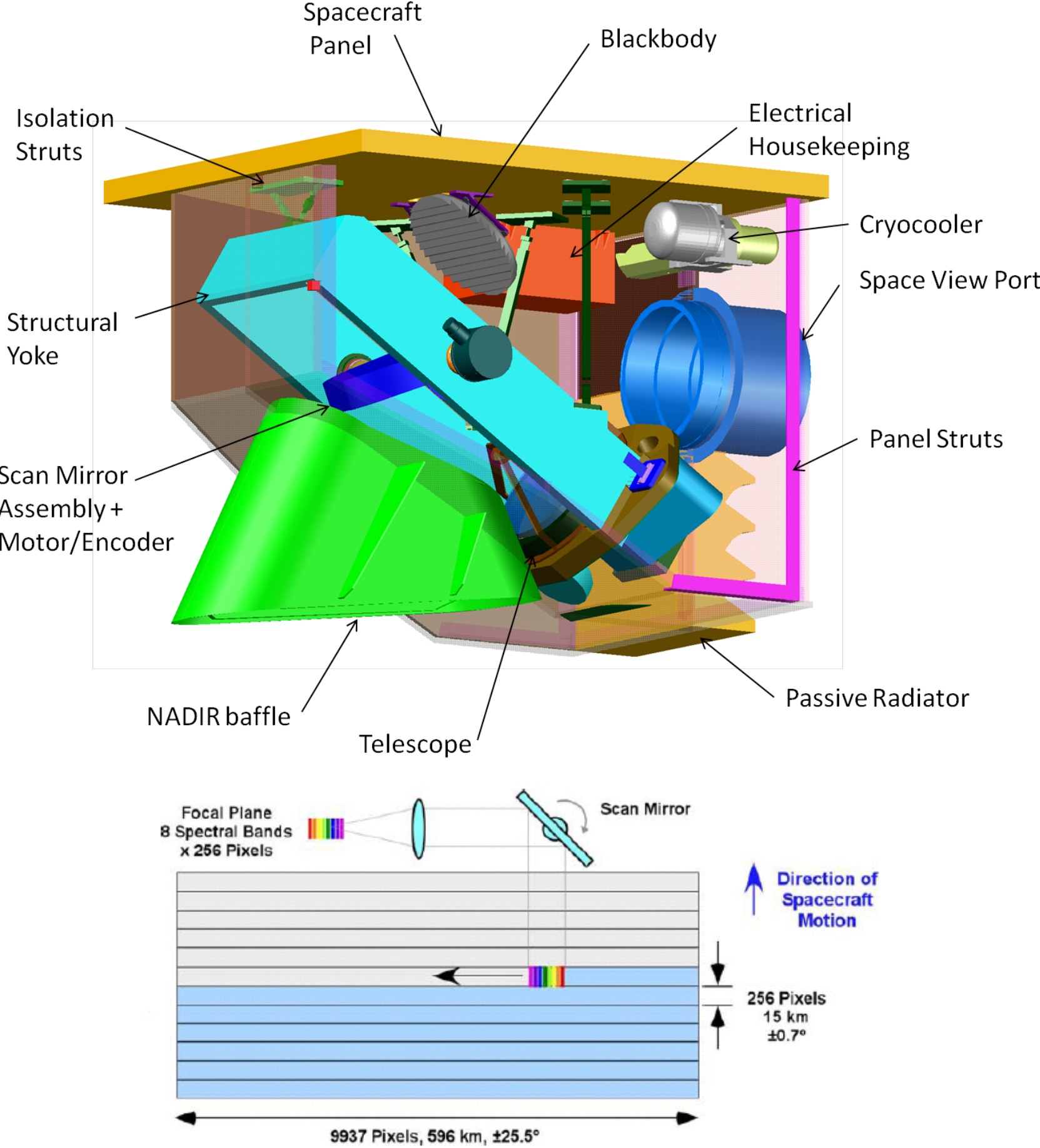
The technology for the HypsIRI-TIR instrument is mature, but further work is needed to reduce risk. In particular, the proposed design requires a high sensitivity and high throughput focal plane array (FPA) coupled with a scanning mechanism which has stringent pointing knowledge. The scanning approach, and the high sensitivity and high throughput FPA, are required to meet the revisit time (5 days), the high spatial resolution (60m), and the number of spectral channels (8) specified by the Decadal Survey and the HypsIRI Science Study Group for the mission. The next step is to reduce the risk associated with the scanning mechanism and the FPA with the development of a laboratory prototype termed the Prototype HypsIRI Thermal Infrared Radiometer (PHyTIR).

- PHyTIR will demonstrate that:
1. The detectors and readouts meet all signal-to-noise and speed specifications.
 2. The scan mirror, together with the structural stability, meets the pointing knowledge requirements.
 3. The long-wavelength channels do not saturate below 480 K.
 4. The cold shielding allows the use of ambient temperature optics on the HypsIRI-TIR instrument without impacting instrument performance.

HypsIRI-TIR Sensor Performance



HypsIRI-TIR Instrument Concept



HypsIRI-TIR Sensor Assembly

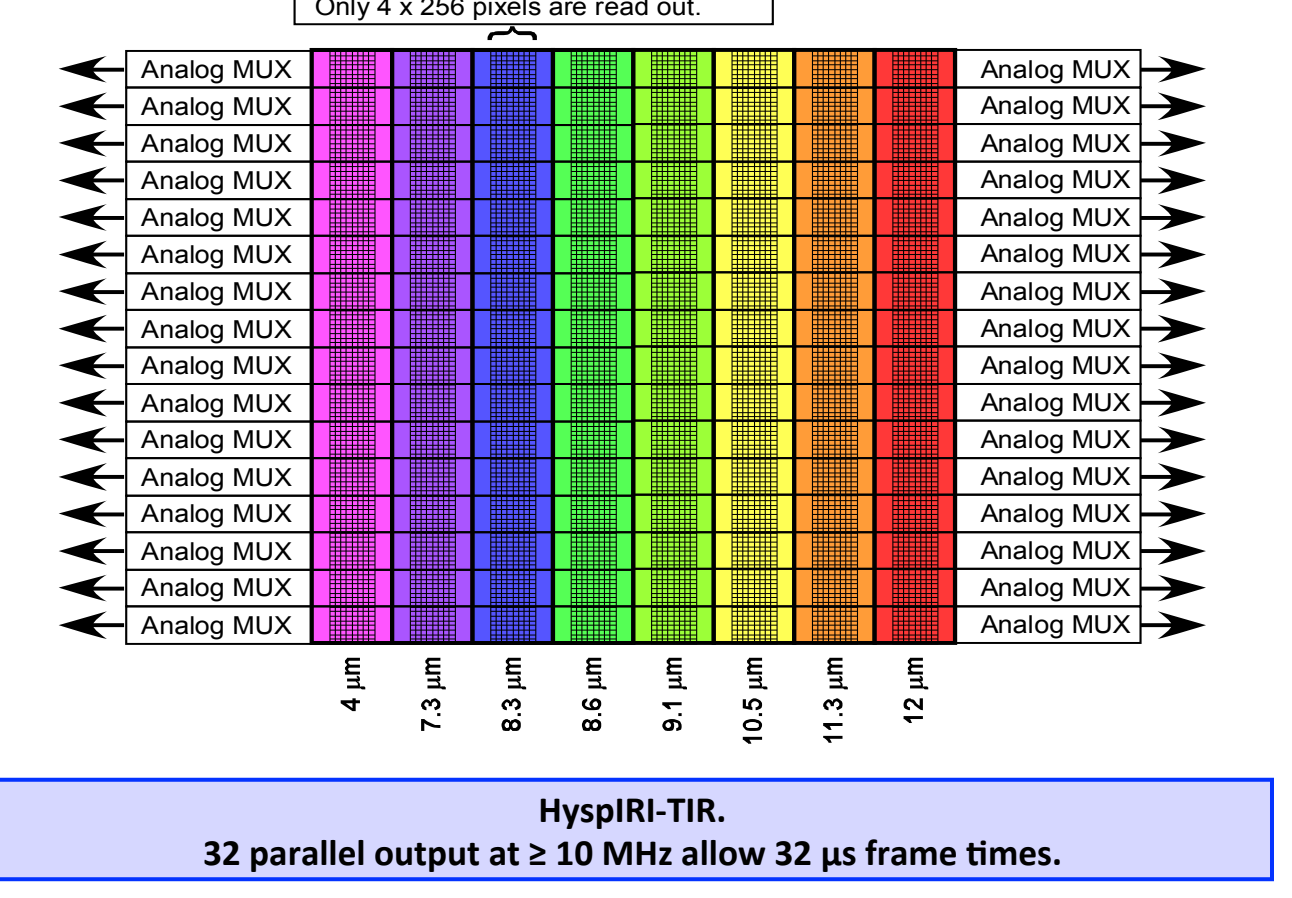
HypsIRI TIR Instrument Characteristics			
Spectral		Spatial	
Bands (8)	3.98 μm , 7.35 μm , 8.28 μm , 8.63 μm , 9.07 μm , 10.53 μm , 11.33 μm , 12.05 μm	IFOV	96 μrad ; 60 m at nadir
Bandwidths	0.084 μm , 0.32 μm , 0.34 μm , 0.35 μm , 0.36 μm , 0.54 μm , 0.54 μm , 0.52 μm	MTF	>0.60 at FNy
Accuracy	<0.01 μm	Scan Type	Push-Whisk
Radiometric		Cross-Whisk Samples	256
Temperature Range	Band 1: 400-1200 K Bands 2-8: 200 K – 500 K	Samples in Whisk Direction (Cross Track)	9,937
Resolution	< 0.05 K, linear quantization to 14 bits	Cross-Whisk Swath Width	15.4 km ($\pm 0.7^\circ$ at 623 km altitude)
Accuracy	< 0.5 K at 250 K	Swath Length in Whisk Direction	596 km ($\pm 25.5^\circ$ at 623 km altitude)
Precision (NETD)	< 0.2 K	Band to Band Co-Registration	0.2 pixels (12 m)
Linearity	>99% characterized to 0.1 %	Pointing Knowledge	10 arcsec (0.5 pixels)

• Butcher-Block Filter Assembly
• Baffles to Prevent Crosstalk Between Spectral Channels

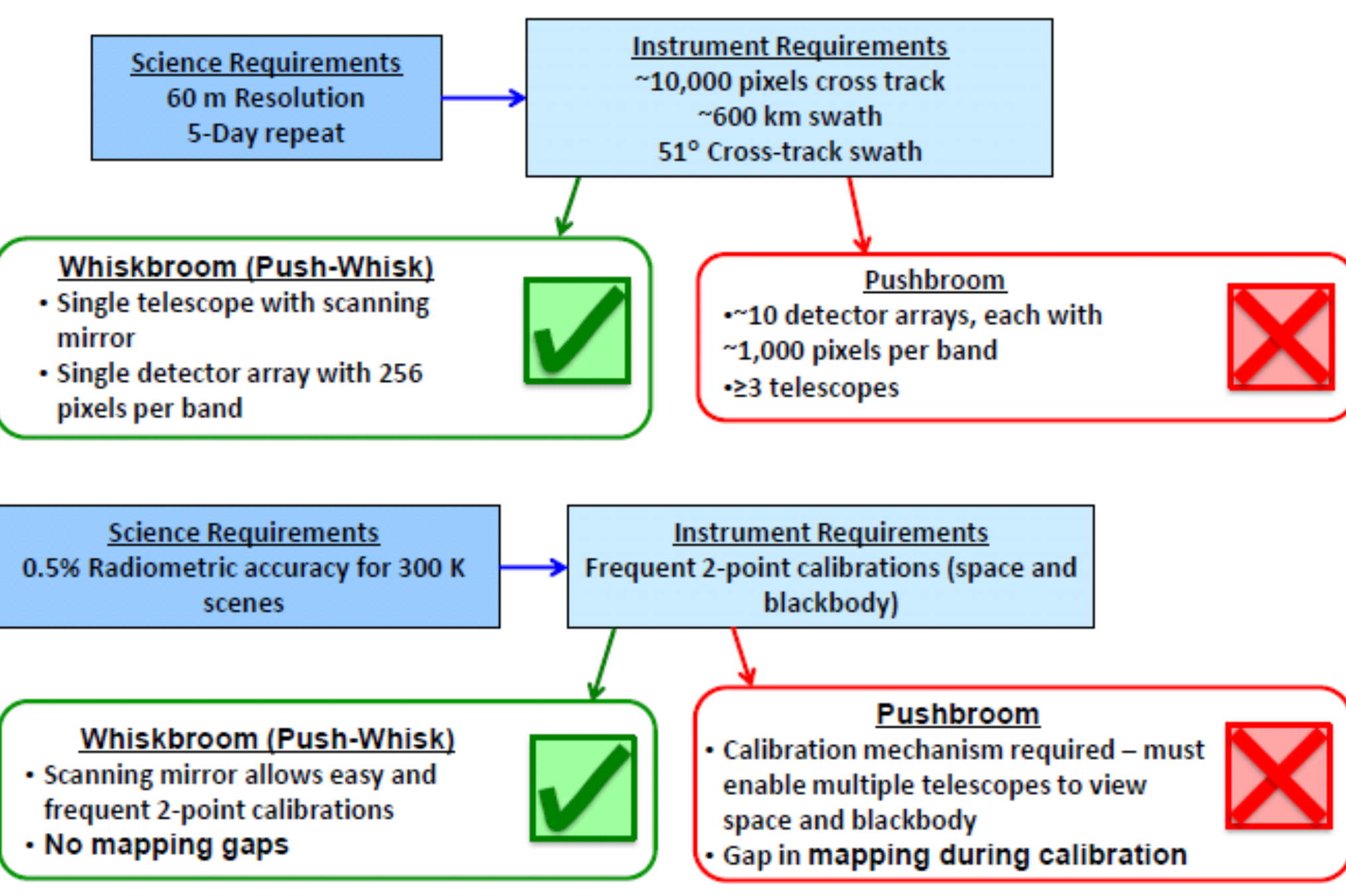
• CMOS Read-Out Integrated Circuit (ROIC)
• Multiple Output Signals to Enable Necessary Pixel Read Rate

• MCT Detector Array – 256 elements cross-sweep
• 1 Bandgap to Cover Full Spectral Range
• ≥ 4 Detector Columns per Spectral Channel to Allow Time Delay and Integration (TDI)

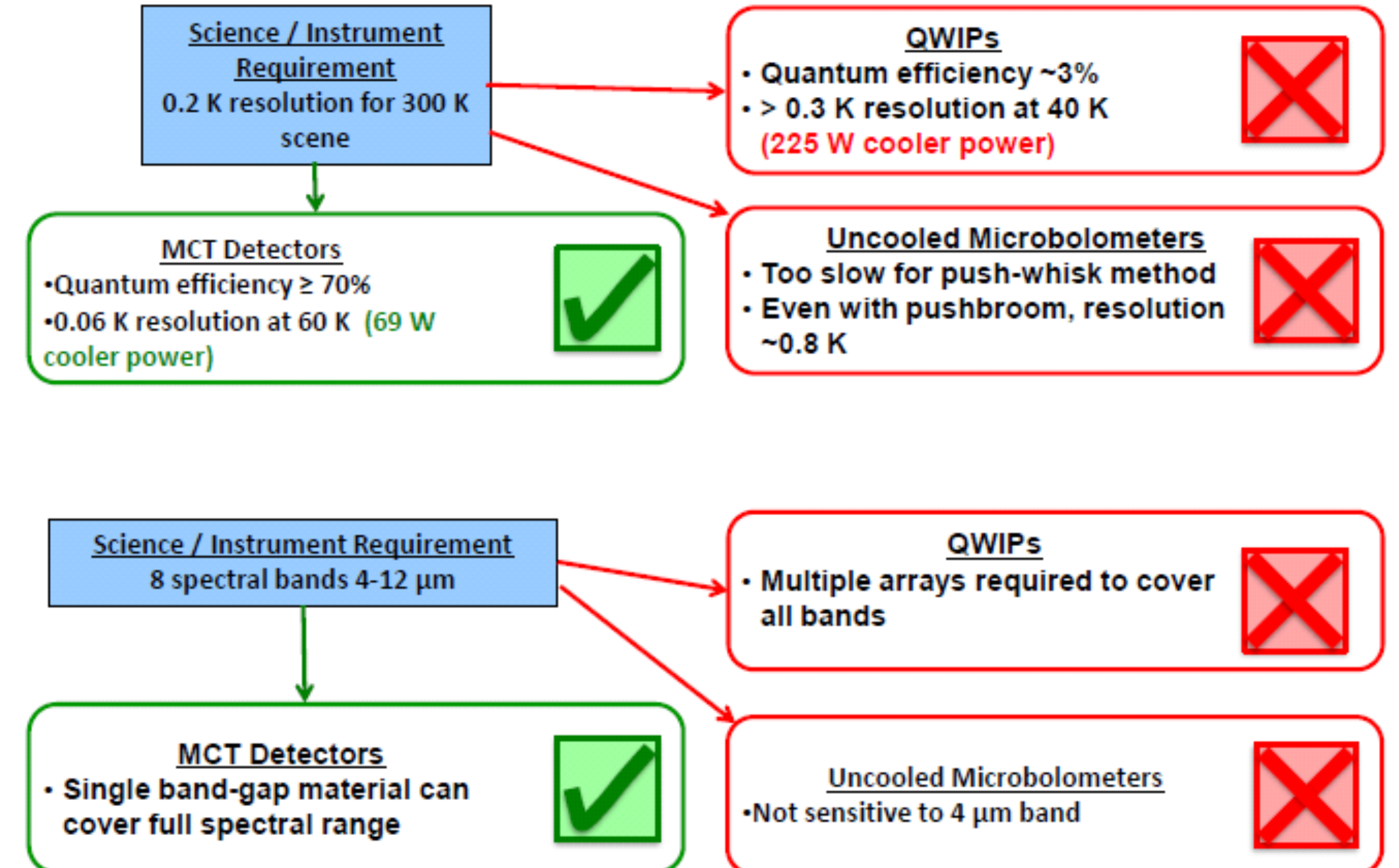
• 60 K Cold Tip of Cryocooler



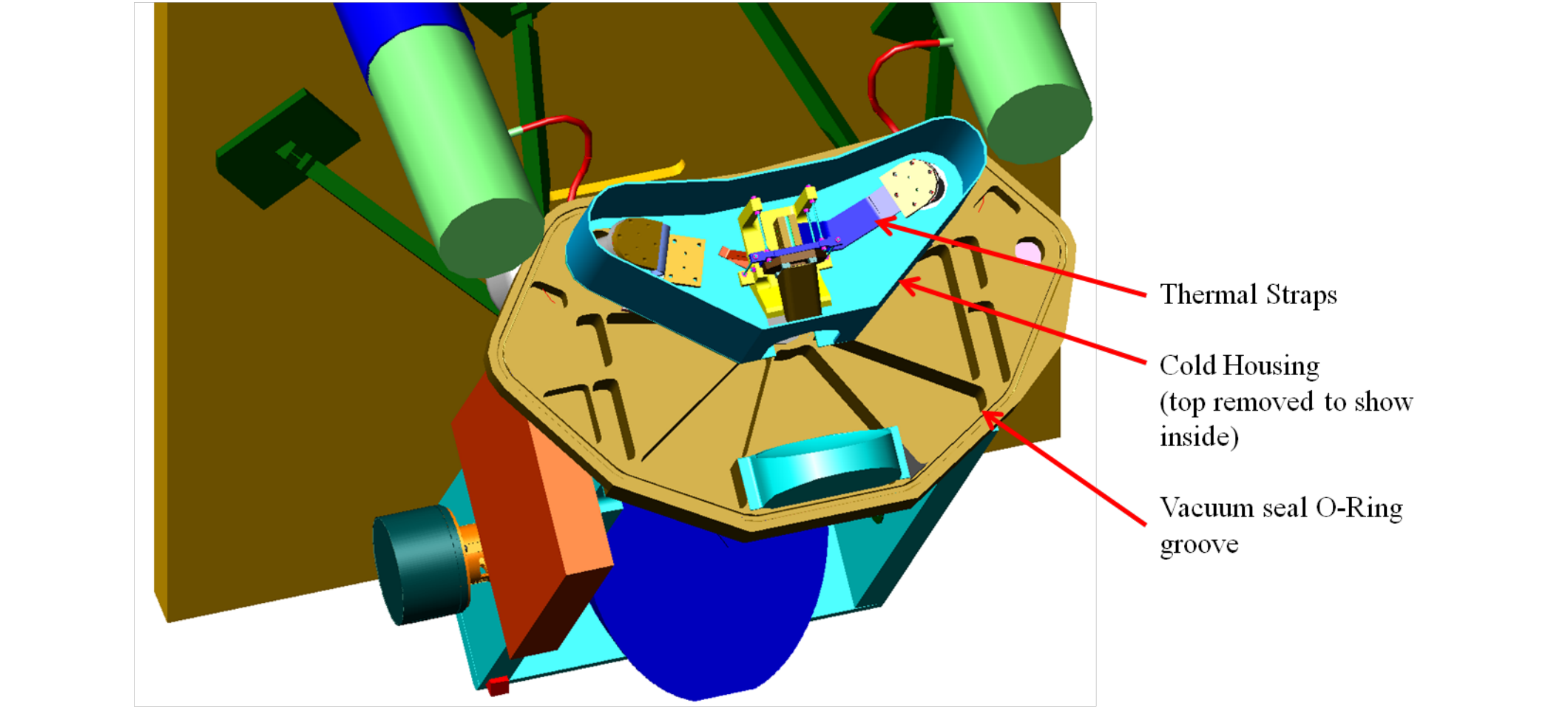
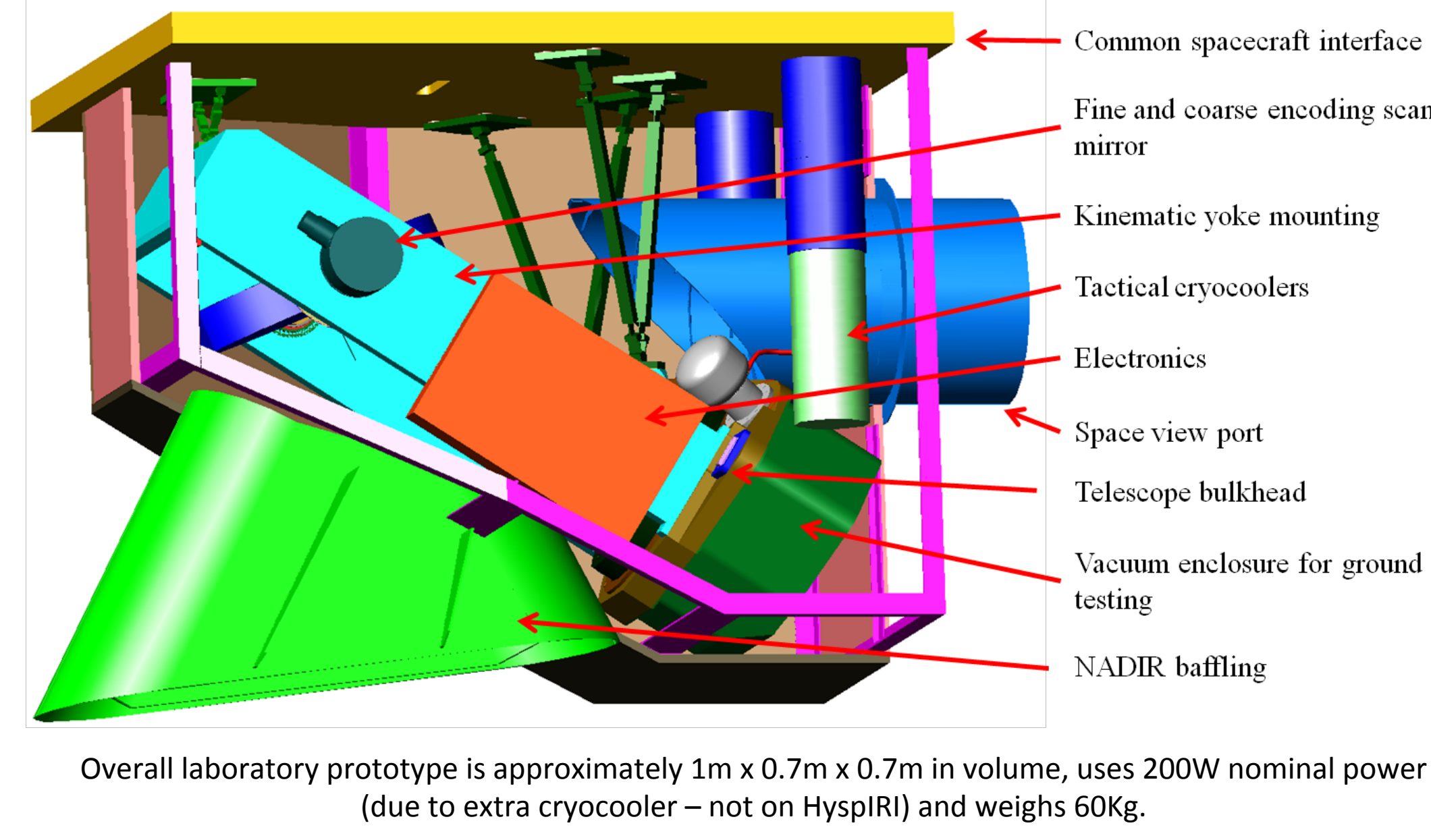
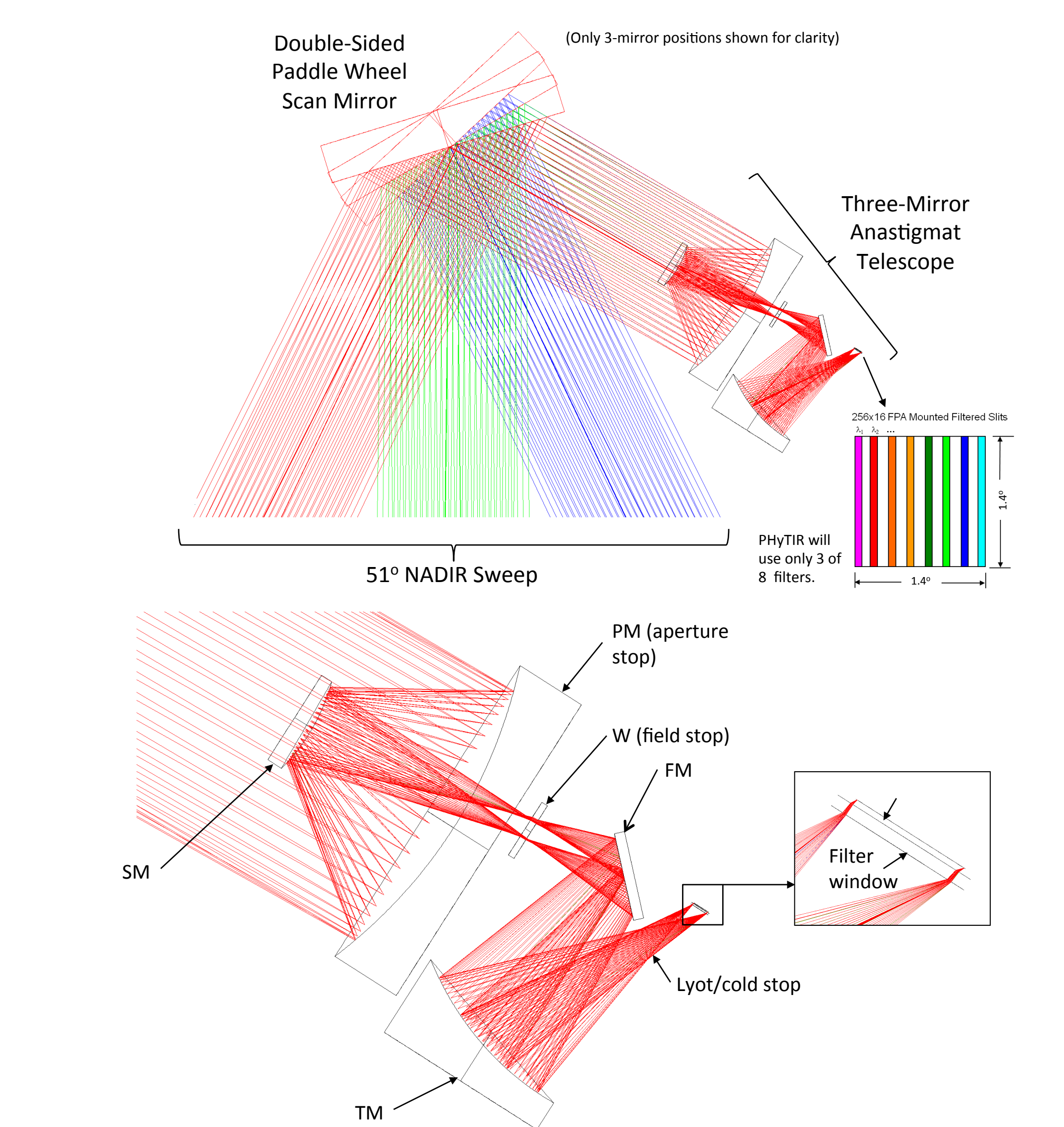
HypsIRI-TIR Sensor Rational



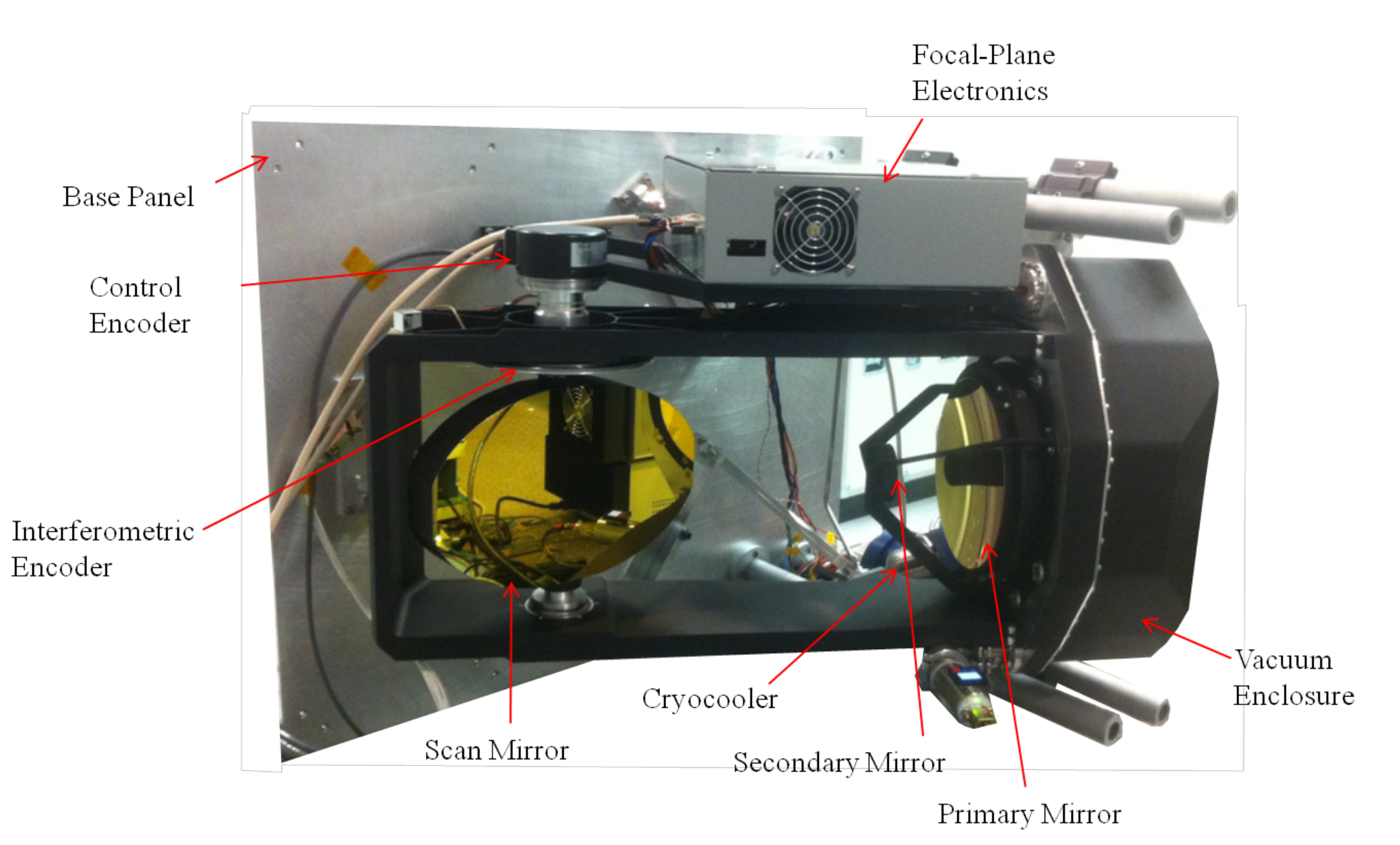
HypsIRI-TIR Sensor Rational



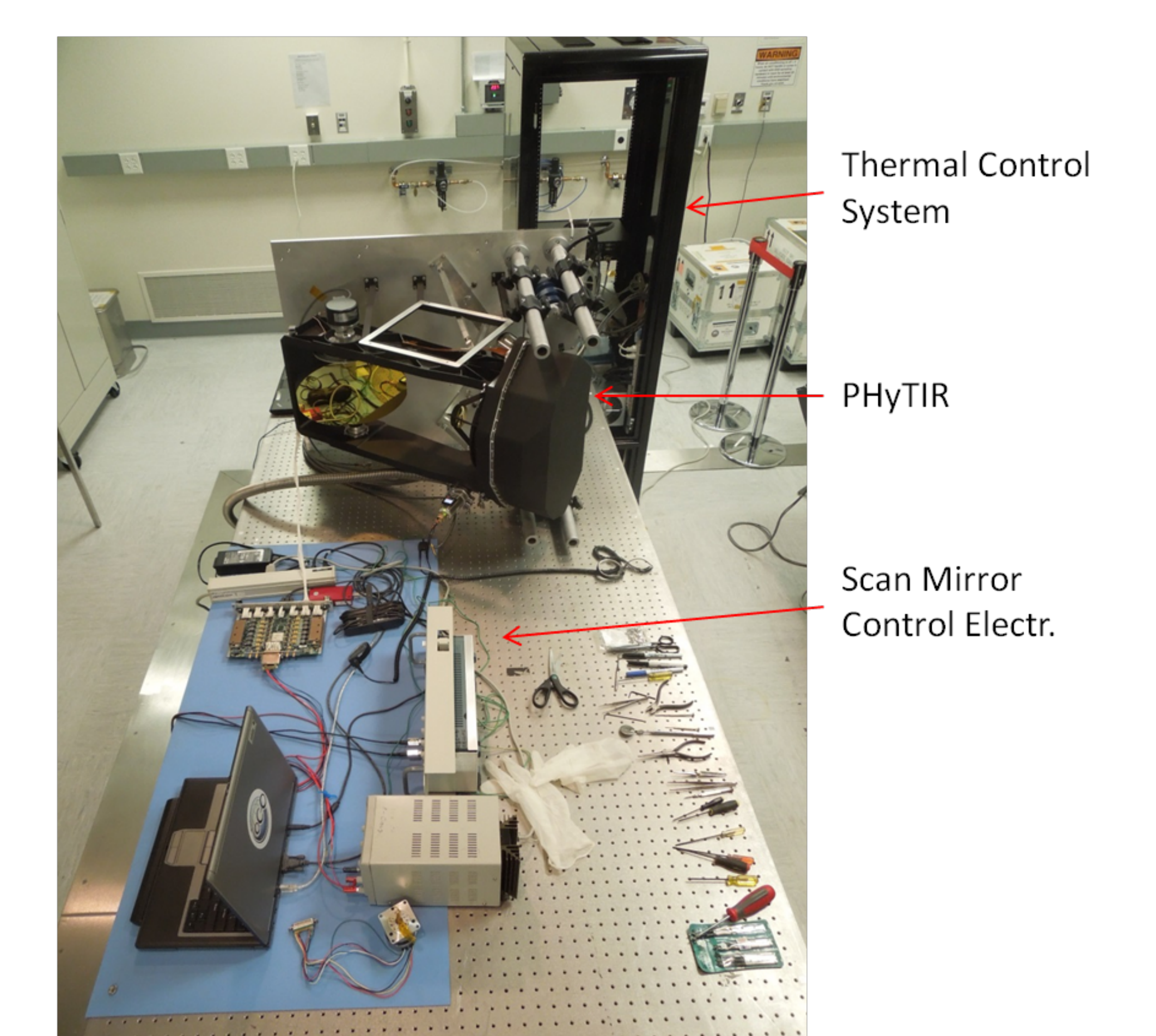
PHyTIR Implementation and Current Design



PHyTIR Build



PHyTIR Calibration and Validation



PHyTIR Summary

PHyTIR will be used to retire four key risks areas as noted earlier. This activity will benefit the development of any airborne or spaceborne system that will utilize a high speed scanning mirror coupled with a MCT detector array to obtain a wide swath width, high spatial resolution, thermal infrared measurement with a NEAT of less than 0.2K.

Similar systems have been used in the Moderate Resolution Imaging Spectroradiometer (MODIS), Visible Infrared Imaging Radiometer Suite (VIIRS), Advanced Spaceborne Thermal Emission Radiometer (ASTER) and Landsat (TM5/ETM+) instruments (Barnes et. al. 1998; Mitchel 2008; Ohmae and Kitamura, 1994; Barsi et al. 2003).

However, none of these existing systems has sufficient performance to meet the measurement requirements of the HypsIRI-TIR instrument. PHyTIR will demonstrate that HypsIRI-TIR required high accuracy measurements can be made and help enable both the HypsIRI-TIR instrument as well as other future instruments built by governments or commercial companies that utilize similar technology.

This research was carried out at the Jet Propulsion Laboratory, California Institute of Technology, under a contract with the National Aeronautics and Space Administration. Reference herein to any specific commercial product, process, or service by trade name, trademark, manufacturer, or otherwise, does not constitute or imply its endorsement by the United States Government or the Jet Propulsion Laboratory, California Institute of Technology.